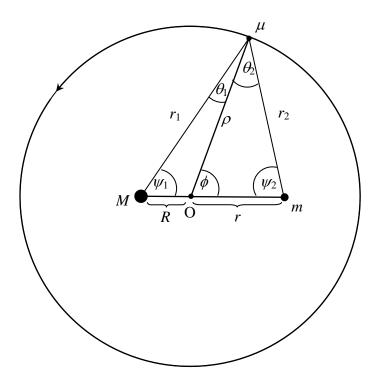


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I. Solution



1.1 Let O be their centre of mass. Hence MR - mr = 0

..... (1)

$$m\omega_0^2 r = \frac{GMm}{\left(R+r\right)^2}$$

$$M\omega_0^2 R = \frac{GMm}{\left(R+r\right)^2}$$
(2)

From Eq. (2), or using reduced mass, $\omega_0^2 = \frac{G(M+m)}{(R+r)^3}$ Hence, $\omega_0^2 = \frac{G(M+m)}{(R+r)^3} = \frac{GM}{r(R+r)^2} = \frac{Gm}{R(R+r)^2}$. (3)



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1.2 Since μ is infinitesimal, it has no gravitational influences on the motion of neither *M* nor *m*. For μ to remain stationary relative to both *M* and *m* we must have:

$$\frac{GM\mu}{r_1^2}\cos\theta_1 + \frac{Gm\mu}{r_2^2}\cos\theta_2 = \mu\omega_0^2\rho = \frac{G(M+m)\mu}{(R+r)^3}\rho \qquad (4)$$

$$\frac{GM\,\mu}{r_1^2}\sin\theta_1 = \frac{Gm\mu}{r_2^2}\sin\theta_2 \qquad (5)$$

Substituting $\frac{GM}{r_1^2}$ from Eq. (5) into Eq. (4), and using the identity $\sin \theta_1 \cos \theta_2 + \cos \theta_1 \sin \theta_2 = \sin(\theta_1 + \theta_2)$, we get

The distances r_2 and ρ , the angles θ_1 and θ_2 are related by two Sine Rule equations

$$\frac{\sin\psi_1}{\rho} = \frac{\sin\theta_1}{R}$$

$$\frac{\sin\psi_1}{r_2} = \frac{\sin(\theta_1 + \theta_2)}{R + r}$$
(7)

Substitute (7) into (6)

$$\frac{1}{r_2^3} = \frac{R}{(R+r)^4} \frac{(M+m)}{m}$$
 (10)

Since $\frac{m}{M+m} = \frac{R}{R+r}$, Eq. (10) gives

$$r_2 = R + r \tag{11}$$

Alternatively,

$$\frac{r_1}{\sin(180^\circ - \phi)} = \frac{R}{\sin\theta_1} \text{ and } \frac{r_2}{\sin\phi} = \frac{r}{\sin\theta_2}$$

$$\frac{\sin\theta_1}{\sin\theta_2} = \frac{R}{r} \times \frac{r_2}{r_1} = \frac{m}{M} \times \frac{r_2}{r_1}$$
Combining with Eq. (5) gives $r_1 = r_2$



Hence, it is an equilateral triangle with

$$\psi_1 = 60^\circ$$
 $\psi_2 = 60^\circ$
(13)

The distance ρ is calculated from the Cosine Rule.

$$\rho^{2} = r^{2} + (R+r)^{2} - 2r(R+r)\cos 60^{\circ}$$

$$\rho = \sqrt{r^{2} + rR + R^{2}}$$
(14)

Alternative Solution to 1.2

Since μ is infinitesimal, it has no gravitational influences on the motion of neither M nor m. For μ to remain stationary relative to both M and m we must have:

$$\frac{GM\,\mu}{r_1^2}\cos\theta_1 + \frac{Gm\mu}{r_2^2}\cos\theta_2 = \mu\omega^2\rho = \frac{G(M+m)\mu}{(R+r)^3}\rho \qquad \dots \qquad (4)$$

Note that

Equations (5) and

 $\psi_1 = \psi_2 \tag{9}$

The equation (4) then becomes:

$$M\cos\theta_1 + m\cos\theta_2 = \frac{(M+m)}{(R+r)^3}r_1^2\rho \qquad (10)$$

Equations (8) and (10):
$$\sin(\theta_1 + \theta_2) = \frac{M+m}{M} \frac{r_1^2 \rho}{(R+r)^3} \sin \theta_2$$
 (11)

Note that from figure,
$$\frac{\rho}{\sin\psi_2} = \frac{r}{\sin\theta_2}$$
 (12)



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Equations (11) and (12):
$$\sin(\theta_1 + \theta_2) = \frac{M+m}{M} \frac{r_1^2 r}{(R+r)^3} \sin \psi_2$$
(13)
Also from figure,
 $(R+r)^2 = r_2^2 - 2r_1 r_2 \cos(\theta_1 + \theta_2) + r_1^2 = 2r_1^2 [1 - \cos(\theta_1 + \theta_2)]$ (14)
Equations (13) and (14): $\sin(\theta_1 + \theta_2) = \frac{\sin \psi_2}{2[1 - \cos(\theta_1 + \theta_2)]}$ (15)
 $\theta_1 + \theta_2 = 180^\circ - \psi_1 - \psi_2 = 180^\circ - 2\psi_2$ (see figure)
 $\therefore \cos \psi_2 = \frac{1}{2}, \ \psi_2 = 60^\circ, \ \psi_1 = 60^\circ$
Hence *M* and *m* from an equilateral triangle of sides $(R+r)$
Distance μ to *M* is $R+r$
Distance μ to *M* is $R+r$
Distance μ to *O* is $\rho = \sqrt{\left(\frac{R+r}{2}-R\right)^2 + \left\{\left(R+r\right)\frac{\sqrt{3}}{2}\right\}^2} = \sqrt{R^2 + Rr + r^2}$

1.3 The energy of the mass μ is given by

$$E = -\frac{GM\mu}{r_1} - \frac{Gm\mu}{r_2} + \frac{1}{2}\mu((\frac{d\rho}{dt})^2 + \rho^2\omega^2)$$
 (15)

Since the perturbation is in the radial direction, angular momentum is conserved ($r_1=r_2=\Re$ and $\ m=M$),

$$E = -\frac{2GM\,\mu}{\Re} + \frac{1}{2}\,\mu \left(\left(\frac{d\,\rho}{dt}\right)^2 + \frac{\rho_0^4\,\omega_0^2}{\rho^2} \right) \tag{16}$$

Since the energy is conserved,

R

60



Since
$$\frac{d\rho}{dt} \neq 0$$
, we have

$$\frac{2GM}{\Re^3}\rho + \frac{d^2\rho}{dt^2} - \frac{\rho_0^4\omega_0^2}{\rho^3} = 0 \text{ or}$$

$$\frac{d^2\rho}{dt^2} = -\frac{2GM}{\Re^3}\rho + \frac{\rho_0^4\omega_0^2}{\rho^3}.$$
(20)

The perturbation from \Re_0 and ρ_0 gives $\Re = \Re_0 \left(1 + \frac{\Delta \Re}{\Re_0}\right)$ and $\rho = \rho_0 \left(1 + \frac{\Delta \rho}{\rho_0}\right)$.

Using binomial expansion $(1 + \varepsilon)^n \approx 1 + n\varepsilon$,

$$\frac{d^{2}\Delta\rho}{dt^{2}} = -\frac{2GM}{\Re_{0}^{3}}\rho_{0}\left(1 + \frac{\Delta\rho}{\rho_{0}}\right)\left(1 - \frac{3\Delta\Re}{\Re_{0}}\right) + \rho_{0}\omega_{0}^{2}\left(1 - \frac{3\Delta\rho}{\rho_{0}}\right).$$
(22)

Using
$$\Delta \rho = \frac{\Re}{\rho} \Delta \Re$$
,
 $\frac{d^2 \Delta \rho}{dt^2} = -\frac{2GM}{\Re_0^3} \rho_0 \left(1 + \frac{\Delta \rho}{\rho_0} - \frac{3\rho_0 \Delta \rho}{\Re_0^2} \right) + \rho_0 \omega_0^2 \left(1 - \frac{3\Delta \rho}{\rho_0} \right).$ (23)
Since $\omega_0^2 = \frac{2GM}{\Re_0^3}$,

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2 \rho_0 \left(\frac{4\Delta\rho}{\rho_0} - \frac{3\rho_0\Delta\rho}{\Re_0^2}\right) \tag{25}$$

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2\Delta\rho \left(4 - \frac{3\rho_0^2}{\Re_0^2}\right) \tag{26}$$

From the figure, $\rho_0 = \Re_0 \cos 30^\circ \text{ or } \frac{{\rho_0}^2}{{\Re_0}^2} = \frac{3}{4}$,

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2\Delta\rho \left(4 - \frac{9}{4}\right) = -\frac{7}{4}\omega_0^2\Delta\rho . \qquad (27)$$



Angular frequency of oscillation is
$$\frac{\sqrt{7}}{2}\omega_0$$
.

Alternative solution:

 $M = m \text{ gives } R = r \text{ and } \omega_0^2 = \frac{G(M+M)}{(R+R)^3} = \frac{GM}{4R^3}. \text{ The unperturbed radial distance of } \mu \text{ is}$ $\sqrt{3}R, \text{ so the perturbed radial distance can be represented by } \sqrt{3}R + \zeta \text{ where } \zeta <<\sqrt{3}R \text{ as}$ shown in the following figure.
Using Newton's 2nd law, $-\frac{2GM\mu}{\{R^2 + (\sqrt{3}R + \zeta)^2\}^{3/2}}(\sqrt{3}R + \zeta) = \mu \frac{d^2}{dt^2}(\sqrt{3}R + \zeta) - \mu \omega^2(\sqrt{3}R + \zeta).$ (1)
The conservation of angular momentum gives $\mu \omega_0(\sqrt{3}R)^2 = \mu \omega(\sqrt{3}R + \zeta)^2$.
(2)
Manipulate (1) and (2) algebraically, applying $\zeta^2 \approx 0$ and binomial approximation. $-\frac{2GM}{\{R^2 + (\sqrt{3}R + \zeta)^2\}^{3/2}}(\sqrt{3}R + \zeta) = \frac{d^2\zeta}{dt^2} - \frac{\omega_0^2\sqrt{3}R}{(1 + \zeta/\sqrt{3}R)^3}$ $-\frac{2GM}{\{4R^3}\sqrt{3}R\frac{(1 + \zeta/\sqrt{3}R)}{(1 + \sqrt{3}\zeta/2R)^{3/2}} = \frac{d^2\zeta}{dt^2} - \frac{\omega_0^2\sqrt{3}R}{(1 + \zeta/\sqrt{3}R)^3}$ $-\frac{GM}{4R^3}\sqrt{3}R\frac{(1 + \zeta/\sqrt{3}R)}{(1 + \sqrt{3}\zeta/2R)^{3/2}} = \frac{d^2\zeta}{dt^2} - \frac{\omega_0^2\sqrt{3}R}{(1 + \zeta/\sqrt{3}R)^3}$

$$-\frac{GM}{4R^{3}}\sqrt{3}R\frac{(1+\zeta/\sqrt{3}R)}{(1+\sqrt{3}\zeta/2R)^{3/2}} = \frac{d^{2}\zeta}{dt^{2}} - \frac{\omega_{0}^{2}\sqrt{3}R}{(1+\zeta/\sqrt{3}R)^{3}}$$
$$-\omega_{0}^{2}\sqrt{3}R\left(1-\frac{3\sqrt{3}\zeta}{4R}\right)\left(1+\frac{\zeta}{\sqrt{3}R}\right) \approx \frac{d^{2}\zeta}{dt^{2}} - \omega_{0}^{2}\sqrt{3}R\left(1-\frac{3\zeta}{\sqrt{3}R}\right)$$
$$\frac{d^{2}}{dt^{2}}\zeta = -\left(\frac{7}{4}\omega_{0}^{2}\right)\zeta$$

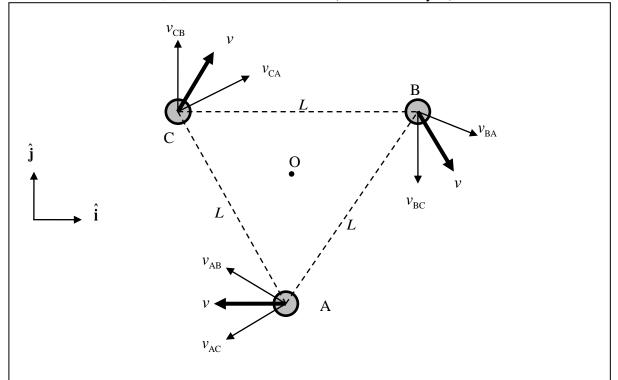
1.4 Relative velocity

Let v = speed of each spacecraft as it moves in circle around the centre O. The relative velocities are denoted by the subscripts A, B and C. For example, v_{BA} is the velocity of B as observed by A.



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The speed is much less than the speed light \rightarrow Galilean transformation.



In Cartesian coordinates, the velocities of B and C (as observed by O) are

For B, $\vec{v}_B = v \cos 60^\circ \hat{\mathbf{i}} - v \sin 60^\circ \hat{\mathbf{j}}$

For C, $\vec{v}_c = v \cos 60^\circ \hat{\mathbf{i}} + v \sin 60^\circ \hat{\mathbf{j}}$

Hence $\vec{v}_{BC} = -2v \sin 60^{\circ} \hat{\mathbf{j}} = -\sqrt{3}v \hat{\mathbf{j}}$ The speed of B as observed by C is $\sqrt{3}v \approx 996$ m/s

Notice that the relative velocities for each pair are anti-parallel.

Alternative solution for 1.4

One can obtain $v_{\rm BC}$ by considering the rotation about the axis at one of the spacecrafts.

$$v_{\rm BC} = \omega L = \frac{2\pi}{365 \times 24 \times 60 \times 60 \text{ s}} (5 \times 10^6 \text{ km}) \approx 996 \text{ m/s}$$

..... (30)